# 4. NORMAL PROCEDURES





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# **SECTION 4**

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#### 4.1 Introduction

Section 4 describes operations and recommended procedures for normal operation of the airplane. Normal procedures following from system installation and optional equipment, which require supplementation of these Instructions, are shown in section 9 - Supplements.

# 4.2 Recommended speeds for normal procedures

#### 4.2.1 Take-off

Climbing speed up to 50 ft (flaps in take-off pos. - 15°) 57 KIAS (66 mph IAS)

Best rate-of-climb speed  $V_Y$  (flaps in take-off pos. - 15°) 57 KIAS (66 mph IAS)

Best rate-of-climb speed  $V_Y$  (flaps retracted - 0°) 65 KIAS (74 mph IAS)

Best angle-of-climb speed  $V_X$  (flaps in take-off pos. - 15°) 54 KIAS (63 mph IAS)

Best angle-of-climb speed  $V_X$  (flaps retracted - 0°) 56 KIAS (65 mph IAS)

#### 4.2.2 Landing

Approaching speed for normal landing (flaps in landing position - 50°) 60 KIAS (69 mph IAS)

# 4.3 Assembly and disassembly

Description of assembly and disassembly is given in the HARMONY LSA Aircraft Maintenance and Inspection Procedures.



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# 4.4 Pre-flight check

Carry out pre-flight check according to the following procedure:

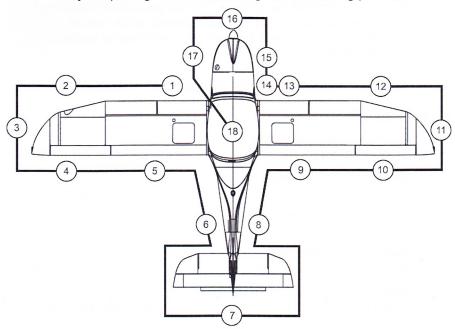


Figure 4-1 Scheme of airplane preflight check

# WARNING CHECK BEFORE PRE-FLIGHT CHECK THAT IGNITION IS SWITCHED OFF!

#### NOTE

The word "condition", used in procedures of preflight check, means visual check of surface, damage, deformation, scratches, attrition, corrosion, icing or other effects decreasing flight safety.





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- 1. Left landing gear leg check
  - landing gear leg attachment and condition
  - · landing gear wheel condition
  - tire condition and inflation
  - · condition and attachment of wheel covers
  - ground cable condition (if installed)
- 2. Left wing check
  - · wing surface condition
  - · leading edge condition
  - · landing light condition if installed
  - · condition of the Pitot tube
  - draining of fuel tank (see Chapter 8, page 8-6)
  - closing of fuel tank cap
- 3. Left wing tip check
  - · surface condition
  - · attachment check
  - fuel tank vent cleanness
  - condition and attachment of the position lights and the anticollision beacon - if installed
- 4. Left aileron check
  - surface condition
  - condition of trim tab (if installed) and its control (electr.trim)
  - attachment
  - free movement
- 5. Left wing flap check
  - surface condition
  - attachment
- 6. Rear part of fuselage check
  - surface condition
  - condition of antennas (top and bottom fuselage surface) if installed





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- 7. Tail units check
  - · tail skid condition
  - surface condition
  - condition of rudder and elevator attachment
  - freedom of rudder and elevator movement
  - condition of trim tab, condition of elevator trim tab control
- 8. Rear part of fuselage check
  - · surface condition
- 9. Right wing flap- see 5.
- 10. Right aileron- see 4. except the trim tab
- 11. Right wing tip see 3.
- 12. Right wing see 2. except the landing light
  - Alternate pitot tube (IFR airplane)
  - AOA probe (if installed)
- 13. Right landing gear leg see 1.
- 14. Front part of the fuselage right hand side check
  - Tip-up canopy attachment and condition
  - condition of the nose landing gear leg
  - nose wheel condition
  - · condition of the nose wheel control rods
  - external power socket (if installed)
- 15. Engine

Checks before the first flight of day - it is necessary to remove upper engine cowling:

- condition of engine bed
- · condition of engine attachment
- · condition of exhaust system
- condition of engine cowlings
- visual check on fuel and electrical system condition





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check on cooling liquid volume in the expansion tank on the engine body (replenish as required up to max. 2/3 of the expansion tank volume)

Checks before every flight:

- cleanness of air intakes
- check on oil level (between marks flattening on the dip
- check on cooling liquid level in the overflow bottle (volume should be approx. 0.42 pints (0.2 litre))
- · proper closing of the upper cowling
- 16. Propeller check
  - attachment
  - · condition of blades, hub and spinner
- 17. Front part of fuselage left hand side check
  - tip-up canopy attachment and condition
- 18. Cockpit check

#### NOTE

Canopy is unlocked if a latch next to lock is visible under the glass, otherwise it is locked. Unlock it first with key.

Master switch

switched on

- Check canopy OPEN/CLOSE indication light (or a message on the EFIS display) function
- all switches

OFF

- instrument equipment
- check on condition
- check of safety belts condition and attachment
- check on presence of loose object in the cockpit
- check on adjusting and securing the rudder pedals (see section 7.3.3) - if installed adjustable rudder pedals

#### **WARNING**

RIGHT AND LEFT PEDAL OF RUDDER CONTROL MUST BE SET TO THE SAME POSITIONS AND WELL SECURED!

AOI and other required documents

check on completness and validity





#### **IRCRAFT OPERATING INSTRUCTIONS**

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# 4.5 Normal procedures and checklist

#### 4.5.1 Before engine starting

Pre-flight check and check on weight and centre of gravity position done
 External power source connect as (if socket is installed) necessary

3. Safety harnesses check, fasten

4. Control stick free5. Rudder pedals free

6. Wing flaps function check
7. Trim tab function check

8. PARKING BRAKE handle (if installed) release brakes

9. Brakes function check

10. AVIONICS SWITCH check OFF11. Ignition check OFF

12. Canopy close

#### 4.5.2 Engine starting

1. Master switch ON

2. Fuel gauge indicators check of fuel qty.

3. FUEL SELECTOR LEFT

Pull the safety button on the fuel selector, turn the handle to the left and then release safety button. Now the handle can be freely moved between left and right position. Safety button prevents unintentionally switch the selector to OFF position.

4. Electric fuel pump (if installed)5. THROTTLE leveridle

6. Choke as necessary (open

by pulling up and lock by turning)

7. Space in the propeller area free





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8. BEACON (if installed)

ON

(if necessary)

9. Brakes

apply

10. Ignition

START (see CAUTION) after starting up BOTH

#### CAUTION

ACTIVATE STARTER FOR 10 SEC. AS A MAXIMUM, AND THEN LET IT COOL DOWN FOR 2 MINUTES.

AFTER STARTING UP ENGINE, DO NOT CARRY OUT SUDDEN RPM CHANGES, AFTER POWER DECREASE WAIT FOR ABOUT 3 S IN ORDER TO REACH CONSTANT RPM BEFORE REACCELERATION.

11. THROTTLE lever

as necessary (see NOTE)

12. Oil pressure

up to 10s min. pressure

13. GEN, AUX GEN (if inst.) switches

#### NOTE

After starting up engine, adjust throttle for smooth engine running at about 2500 RPM. Check oil pressure. Pressure must increase within 10 s. Increase engine RPM until oil pressure is stabilized over 2 bar (29 PSI).

14. Engine instruments

check

15. Choke

as necessary

16. Engine warming up

see NOTE

#### NOTE

Begin warming up with engine running at 2000 RPM. for about 2 minutes, continue at 2500 RPM. Warming time depends on outside air temperature until oil temperature reaches 122 °F.

17. FUEL SELECTOR

RIGHT

Verify proper engine feeding from the right tank for approx. 1 minute.

18. FUEL SELECTOR

LEFT





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NOTE

Start engine with the fuel selector set to **LEFT**. If you would start the engine with the fuel selector set to **RIGHT** and the left tank is full, than fuel bleed from the left tank vent may occur (and pollute environment) because a fuel return hose is led only into the left tank and returning fuel will overfill the left tank.

<ol><li>External power source (if socket is installed)</li></ol>	if used, give instruction to disconnect it
20. AVIONICS SWITCH	ON
21 Radiostation/avionics	ON

22. Other electrical equipment ON as necessary

4.5.3 Before taxiing

Transponder (if installed)
 Outside lights (if installed)
 as necessary

4.5.4 Taxiing

THROTTLE lever as necessary
 Brakes check by depressing
 Rudder pedals function check

4. Direction of taxiing control by rudder pedals (these are mechanically connected with nose wheel control), possibly by slacking up left and right wheel of the main landing gear.





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#### 4.5.5 Before take-off

1. Brakes

brake

2. Ignition check

carry out, see NOTE

#### NOTE

Carry out ignition check in the following way: Set engine speed to 4000 RPM. Switch ignition gradually to L, BOTH, R position and return to BOTH.

RPM drop with one ignition circuit switched off must not exceed 300 RPM. Maximum RPM difference at using one of the L or R circuits is 120 RPM.

3. Engine instruments

check

4. Control stick

free

5. Wing flaps

Take-off pos. (15°)

6. Elevator trim

**NEUTRAL** 

7. Aileron trim (if installed)

**NEUTRAL** 

8. Fuel gauge indicator

check on fuel quantity

8. FUEL SELECTOR

check LEFT

10. CARB.R PREHEATER (if installed)

check function

then OFF

#### NOTE

If CARBURETTOR PREHEATER is switched ON, then engine RPM drop reaches approximately 50 RPM

11. Engine instruments

check

12. Flight instruments

check

13. Radiostation / avionics

check, set

14. Ignition

15. Choke

check BOTH

close (in inserted position)

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16. Master switch check ON

17. Safety harnesses tighten up

18. Canopy closed

19. Transponder (if installed) ON or ALT

#### 4.5.6 Take-off

1. THROTTLE lever max. take-off power

2. During take-off run smoothly lighten up the nose landing gear until airplane take-off occurs.

3. Airspeed

57 KIAS (66 mph IAS)

4. Brakes

brake to stop main

wheel rotation

5. After reaching 150 ft, set flaps to retracted pos. (0°)

6. Trim

as necessary

# WARNING TAKE-OFF IS PROHIBITED:

**× IF ENGINE RUNNING IS IRREGULAR** 

**× IF CHOKE IS OPEN** 

\* IF VALUES OF ENGINE INSTRUMENTS ARE NOT WITHIN THE REQUIRED RANGE

#### 4.5.7 Climb

1. THROTTLE lever

max. continuous power

2. Airspeed

 $V_Y$  = 65 KIAS (75 mph IAS) for the best rate of climb or  $V_X$  = 56 KIAS (64 mph IAS)

for the best angle of climb

3. Engine instruments

check

4. Trim

as necessary

5. Electric fuel pump (if installed) OFF





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#### 4.5.8 Cruise

1. THROTTLE lever

as necessary

2. Airspeed

max. 5500 RPM

3. Engine instruments

check

4. Fuel quantity

check

#### **CAUTION**

**GAUGES** DISPLAY TRUE **FUEL** QUANTITY ONLY ON GROUND AND IN A LEVEL FLIGHT. TO READ TRUE FUEL TRANSITION QUANTITY **AFTER** CLIMB/DESCENT WAIT APPROX. 2 MINUTES TO FUEL TO LEVEL.

#### NOTE

It is recommended to alternately switch the tanks during cruise to equally consume fuel from both tanks and minimize airplane tendency to bank with unbalanced tanks.

Do not fly with the fuel selector set to RIGHT if the left tank is full to avoid fuel bleed from left tank vent.

When the left tank fuel gauge indicates approx. 1/8 of fuel quantity (needle in the middle between 1/4 and 0) then switch to the right tank to consume remaining fuel and then switch back the left tank to complete the flight at left tank. If the engine conks out due to fuel consumption from either tank, then immediately switch the fuel selector to other tank and engine run will be recovered within 7 seconds.

5. CARB.PREHEATER (if installed) as necessary





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#### 4.5.9 Descent

THROTTLE lever as necessary
 Airspeed as necessary
 Trim as necessary
 Engine instruments check

5. CARB.PREHEATER (if installed) as necessary

#### CAUTION

AT LONG APPROACHING AND DESCENDING FROM HIGH ALTITUDE IT IS NOT SUITABLE TO REDUCE THROTTLE TO MINIMUM FOR THE REASON OF POSSIBLE ENGINE UNDERCOOLING AND SUBSEQUENT LOSS OF POWER. PERFORM DESCENDING AT INCREASED IDLE AND CHECK OBSERVANCE OF THE ALLOWED VALUES ON ENGINE INSTRUMENTS.

#### 4.5.10 Before landing

1. Fuel quantity

check

### **CAUTION**

FUEL GAUGES DISPLAY TRUE FUEL QUANTITY ONLY ON GROUND AND IN A LEVEL FLIGHT. TO READ TRUE FUEL QUANTITY AFTER TRANSITION FROM CLIMB/DESCENT WAIT APPROX. 2 MINUTES TO FUEL TO LEVEL.

2.	FUEL SELECTOR	LEFT
3.	Engine instruments	check
4.	Brakes	check by depressing pedals
5.	Safety harnesses	tighten up
6.	Free area of landing	check
7.	CARB.PREHEATER (if installed)	ON





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8. Approaching speed

60 KIAS (69 mph IAS)

9. Flaps

Take-off pos. (15°)

o. 1 lap

10. Trim

as necessary

11. Parking brake (if installed)

check for lever down

#### **CAUTION**

PARKING BRAKE MUST BE RELEASED (LEVER DOWN) TO PREVENT LANDING WITH BRAKED WHEELS.

12. Electric fuel pump (if installed)

ON

#### 4.5.11 FINAL

1. Flaps

landing pos. (30° or 50°)

2. Maintain airspeed

60 KIAS (69 mph IAS)

3. Trim

as necessary

4. CARB.PREHEATER (if installed)

OFF

#### 4.5.12 Balked landing

1. THROTTLE lever

max. take-off power

2. Flaps

take-off pos. (15°)

3. Airspeed

56 KIAS (65 mph IAS)

4. Flaps in 150 ft

retracted pos. (0°)

5. Trim

as necessary

6. THROTTLE lever

max. continuous power

7. Instruments

check

8. Climb at airspeed

65 KIAS (74 mph IAS)

#### 4.5.13 Landing

1. THROTTLE lever

idle

2. Touch-down on main landing gear wheels

carry out

3. Brakes after nose landing gear wheel touch-down

as necessary

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# 4.5.14 After landing

1.	Flaps	retracted pos. (0°)
2.	Trim	NEUTRAL
3.	Outside lights (if installed)	OFF
4.	Transponder (if installed)	OFF
5.	Electric fuel pump (if installed)	OFF

# 4.5.15 Engine shut-off

1.	THROTTLE lever	idle
2.	Engine instruments	check
3.	AVIONICS SWITCH	OFF
4.	Radiostation / avionics	OFF
5.	Other electrical equipment	OFF
6.	Ignition	OFF
7.	BEACON (if installed)	OFF
8.	Master switch	OFF





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#### 4.5.16 Airplane parking

1. Ignition

check OFF

2. Master switch

check OFF

3. FUEL SELECTOR

OFF

Pull the safety button on the fuel selector, turn the handle to the OFF position and then release safety button. Now the handle is blocked in the OFF position. Safety button prevents unintentionally switch the selector from the OFF position.

4. PARKING BRAKE handle (if installed) brake as necessary

5. Canopy

close,

lock as necessary

#### NOTE

It is recommended to use parking brake (if installed) for short-time parking only, between flights during a flight day. After ending the flight day or at low temperatures of ambient air, do not use parking brake, but use the wheel chocks instead.

If after releasing the parking brake wheels (or one of them) stay blocked, depress and release the brake pedals. This procedure should result to unblock wheels. If the wheel or wheels fail to unblock by the process (even after repeated depressing) it is necessary to allow vent valve (in the appropriate wheel) and reduce the pressure in the brake circuit.





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